



Aerospace Data Storage and Processing Systems

Memory Technology for Space

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Motivation

- ❑ When considering PLD performance, peripheral devices must be included
- ❑ After the processor, memory is often *the* device limiting space system capability



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Overview



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❑ Memory types covered

- Non-volatile

- EEPROM

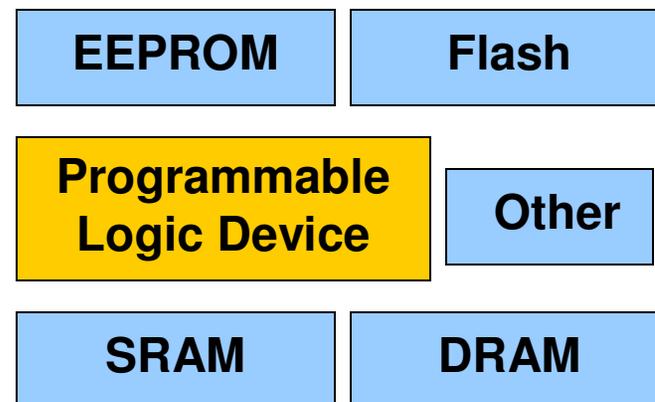
- Flash

- Other

- Volatile

- SRAM

- DRAM



❑ Other special types are used as well

❑ All results from the open literature and “ITAR safe”

❑ Internal PLD memory not included in this discussion

Memory Specialization

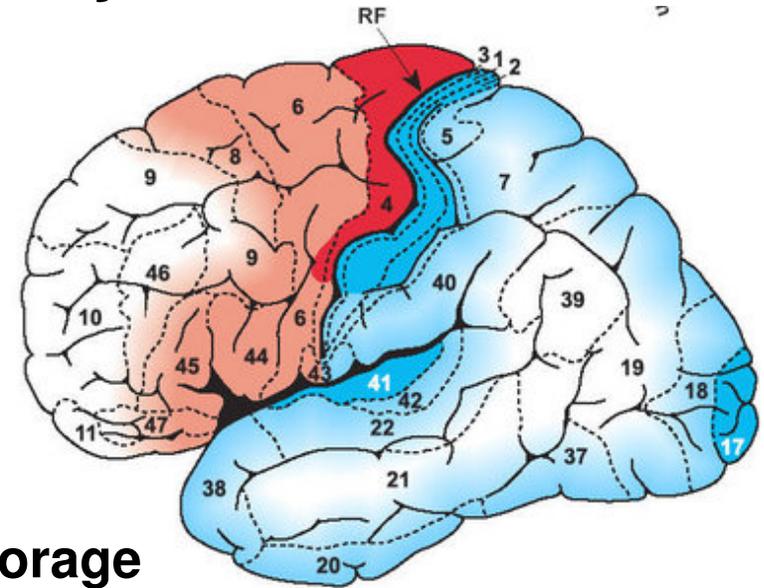
□ Specialization seen in commercial systems also prevalent in space systems

□ Non-volatile memory

- Persistent values, also...
- Greater capacity
- Lower power dissipation
- Slower access times, throughput
- Apps: start-up memory, persistent storage

□ Volatile memory

- Values are not persistent, also ...
- Decreased capacity
- Increased power dissipation
- Fast access times, throughput
- Apps: run-time memory, buffering



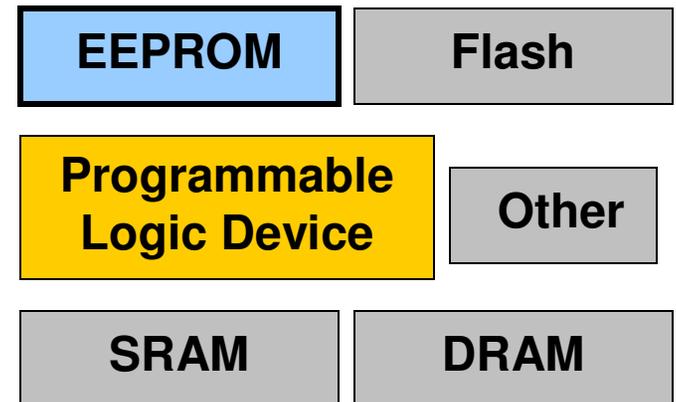
c/o Scholarpedia.com

Non-volatile: EEPROM



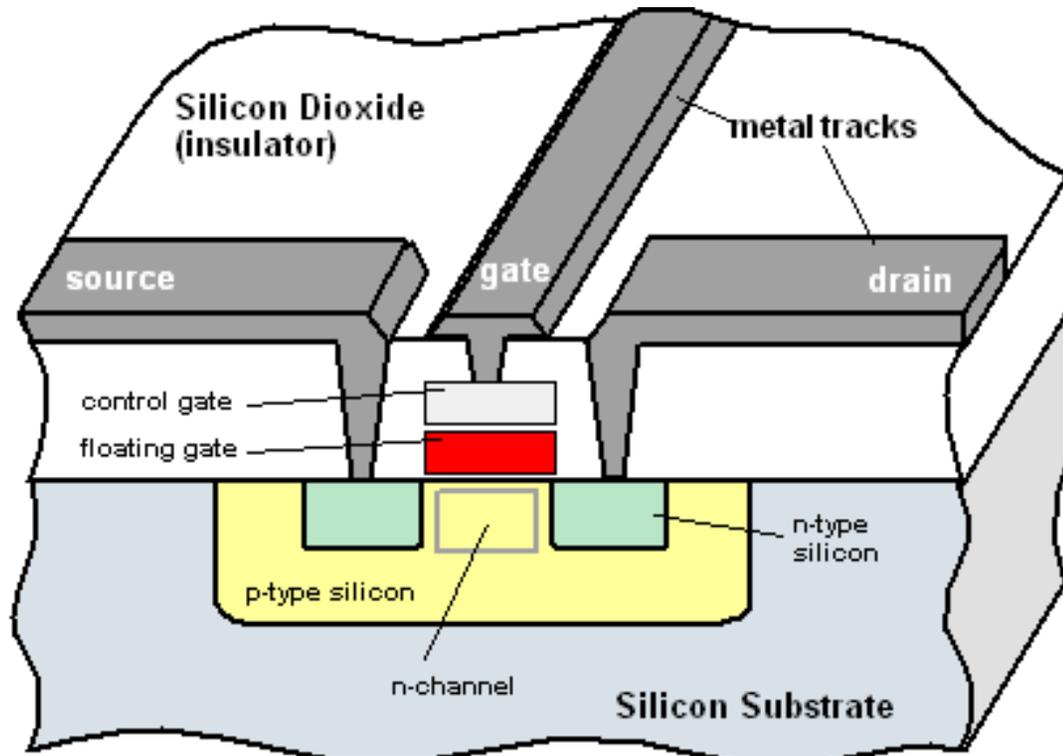
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- ❑ **Electrically Erasable Programmable Read-Only Memory**
- ❑ **Typically used to store small amounts of data for startup**
 - Device configuration
 - Calibration tables
 - Boot code
 - Debug information
- ❑ **Typical device capabilities include***
 - 32Kb to 256Kb options
 - ~1,000,000 rewrite cycles
 - ~10-year data retention or more
 - 10K to 1M rad TID tolerance
- ❑ **Options: Actel, Aeroflex, Atmel, Hitachi, Infineon, Maxwell, Samsung, etc.**



**Data taken from open literature and company websites*

EEPROM Structure



c/o The Computer Language Company

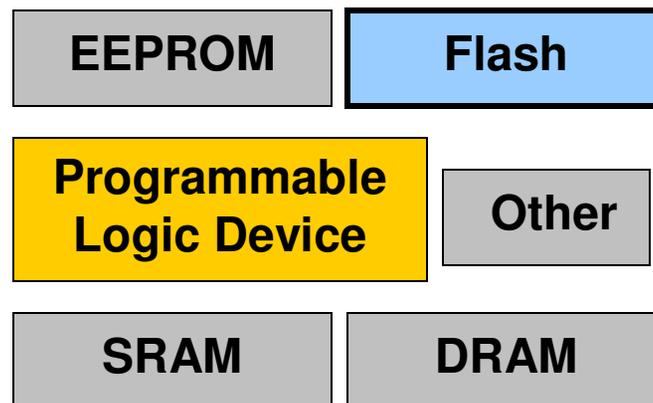
- ❑ Arrays of floating gate transistors provide data persistence and relatively strong TID and SEE performance
- ❑ Control circuitry, especially charge pumps, susceptible

Non Volatile: Flash



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- ❑ Special type of EEPROM that offers fast erase
- ❑ NAND and NOR varieties
- ❑ Typically used to store large amounts of data for startup
 - Device configuration
 - Calibration tables
 - Boot code
 - Debug information
- ❑ Typical device capabilities include*
 - 256Mb to 8Gb options typical
 - ~5,000 to ~500,000 rewrite cycles
 - ~20-year data retention or more
 - 5K to hundreds of K rad TID tolerance typical (ELDRS effects)
- ❑ Options: Hynix, Intel, Micron, Samsung, Spansion, etc.
- ❑ Xilinx offers flash-based configuration memory as well

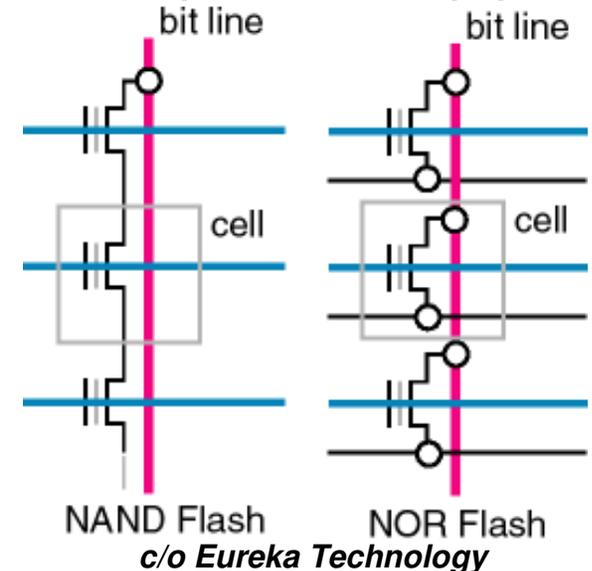


Flash Structure



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- ❑ Arrays of floating gate transistors
- ❑ Different interfaces used in two options
 - NOR: random-access, NAND: page access
- ❑ NOR optimized for speed
 - Ideal for lower-density, high-speed read applications such as code-boot
- ❑ NAND optimized for small cell size
 - Random access traded for lower cost-per-bit and faster write/erase via block operations
 - Ideal for high-density, high-speed program and erase applications, i.e. data-storage
- ❑ Market driven by NAND applications
 - USB drives and iPhone/iPod
 - We are “safe” as long as interests overlap



c/o mediahaven.com



c/o Apple

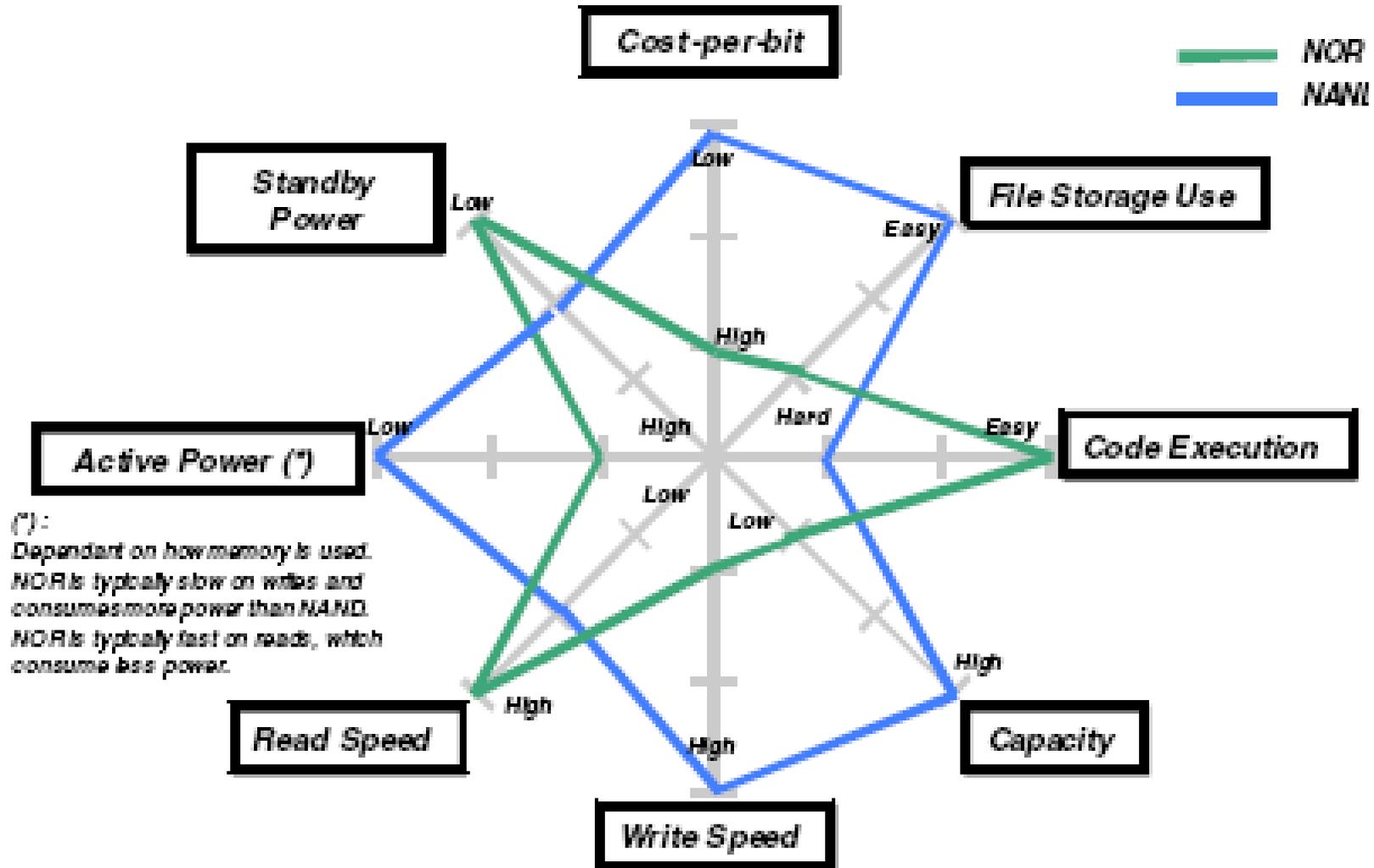


c/o Apple

NOR versus NAND Flash



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c/o Toshiba [2]

Flash Radiation Performance



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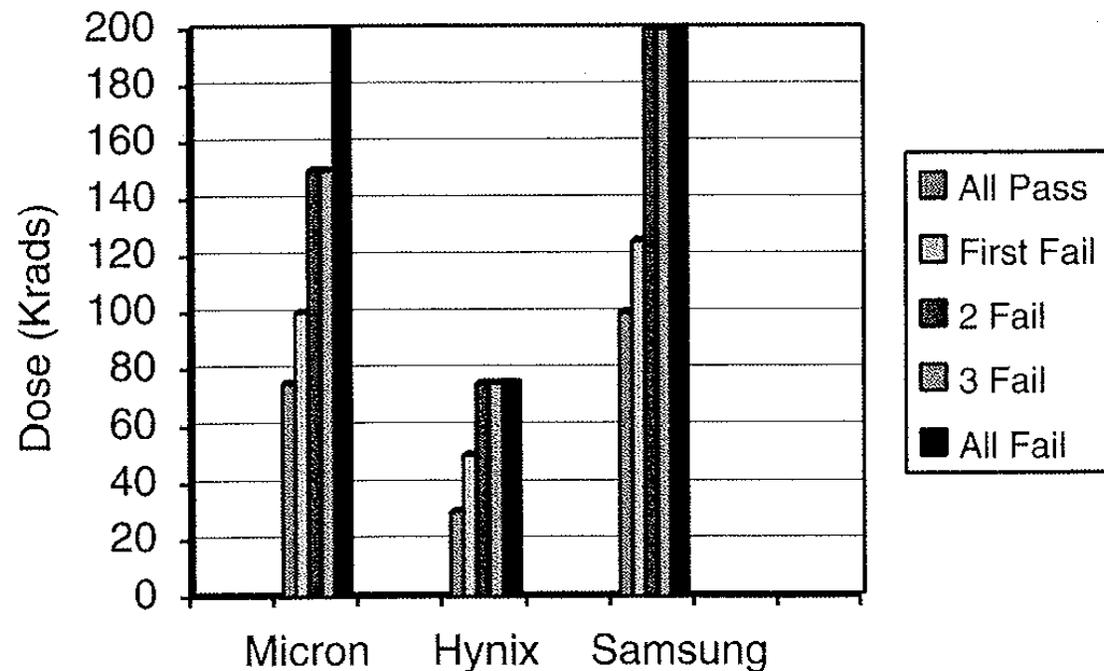


Fig 1. TID response for Micron, Hynix, and Samsung 4G NAND flash memories. *c/o IEEE and Oldham et al. [3]*

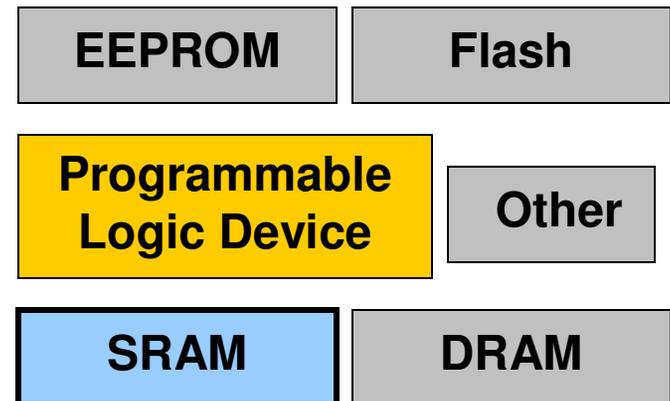
- ❑ Results reprinted from Oldham et al. [3] (NASA-NEPP/DTRA flash study)
- ❑ Static read results shown, other dynamic tests – devices up to 700Krad
- ❑ TID performance improving, due smaller feature size reduction “tricks”
- ❑ SEE affected by Multi-level Cell (MLC) trend and feature size reduction

Volatile: SRAM



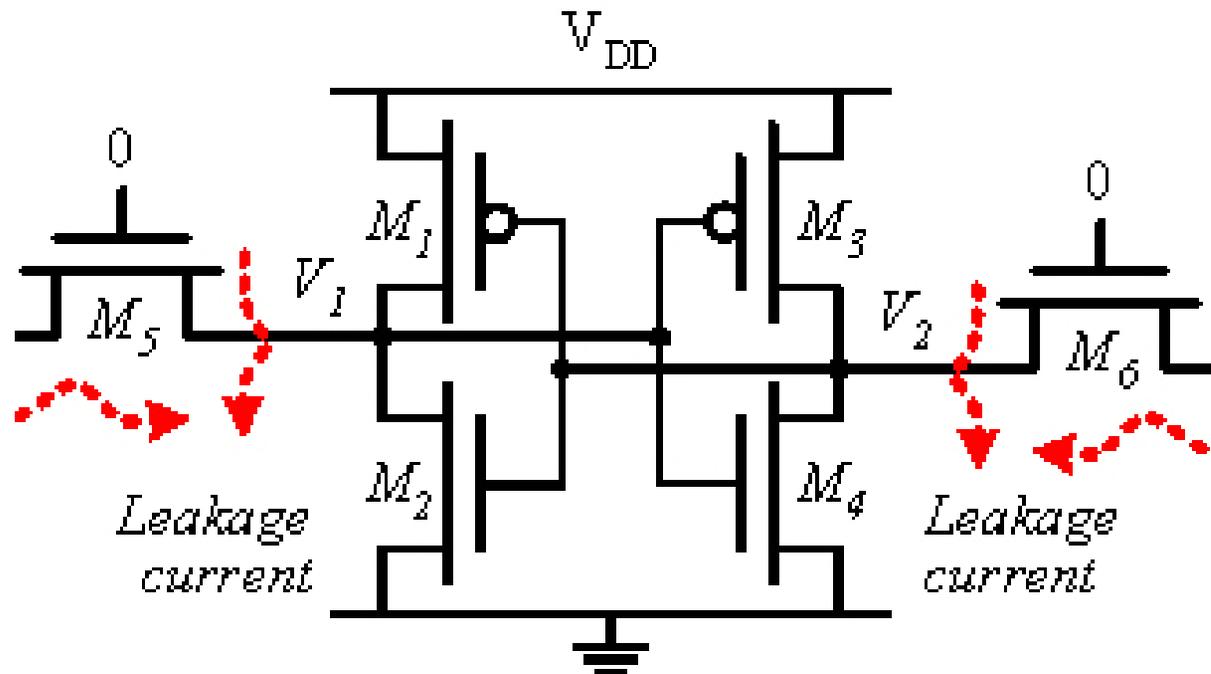
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- ❑ **Static Random Access Memory**
- ❑ **Typically used for control processing applications**
 - Buffer between cache and storage
 - Processor's "main memory"
- ❑ **Typical device capabilities include***
 - 4Mb to 64Mb options typical
 - "infinite" rewrite cycles
 - No data retention
 - 100K to 1M rad TID tolerance typical
 - ~10ns access latency typical
- ❑ **Options: Aeroflex, BAE, Honeywell, Maxwell, Samsung, etc.**



**Data taken from open literature and company websites*

SRAM Structure



c/o Huifang Qin at UC Berkeley (lab web page)

- ❑ Cell uses bi-stable latching circuitry for bit storage
- ❑ Variations of the 6T cross-coupled inverters with buffers
- ❑ Self-reinforcing nature improves SEE performance
- ❑ SEFI modes not as varied and often not as complex

Volatile: DRAM



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❑ Dynamic Random Access Memory

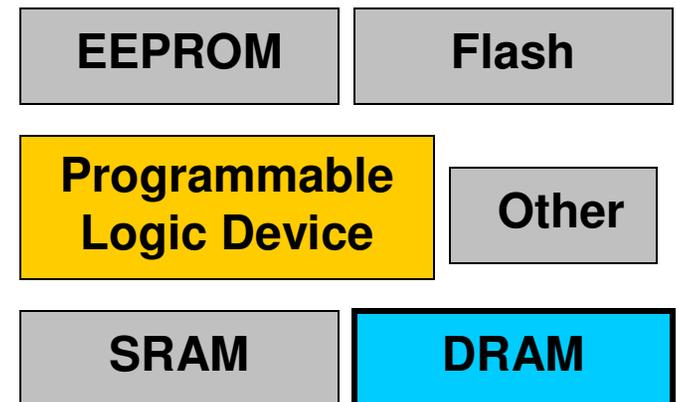
- Synchronous (S), Double Data Rate (DDR) options typical
- e.g. DDR2 SDRAM

❑ Typically used to data processing applications

- Buffer between cache and storage
- Processor's "main memory"

❑ Typical device capabilities include*

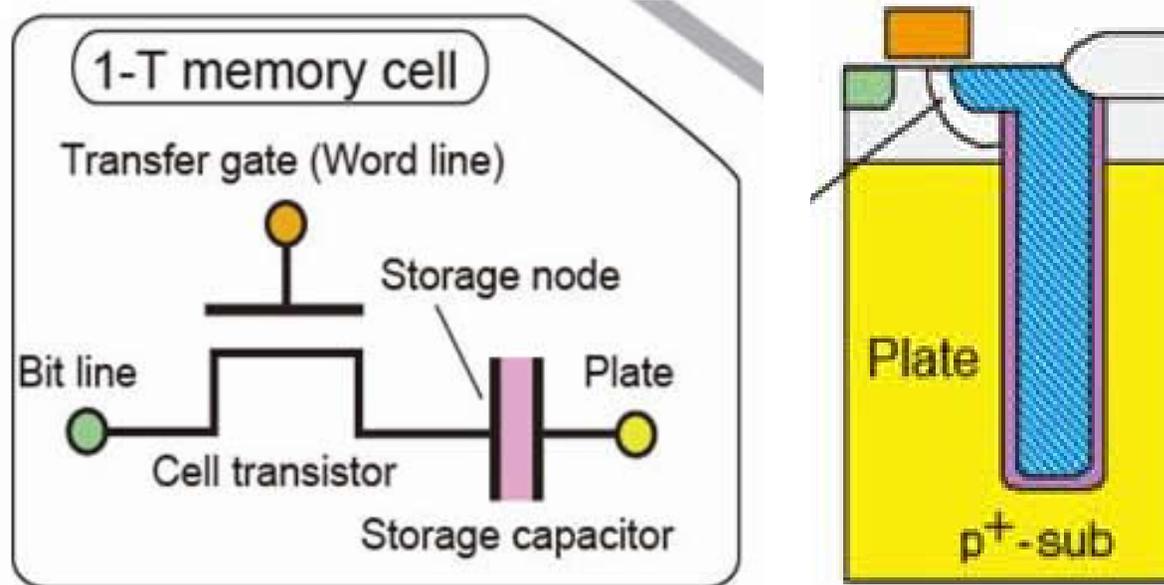
- 2Gb to 8Gb options typical
- "infinite" rewrite cycles
- No data retention
- Variable TID tolerance
- ~50ns access latency typical



❑ Options: Hyundai, Micron, Samsung, etc. (commercial)

**Data taken from open literature and company websites*

DRAM Structure



c/o IEEE.org

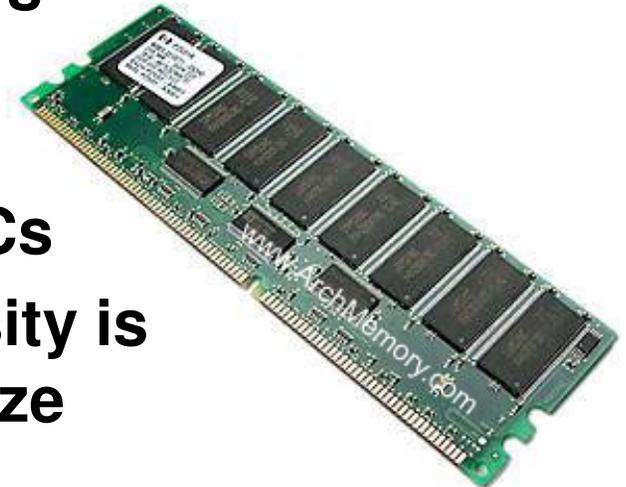
- ❑ Cell uses pass transistor and capacitor for bit storage
- ❑ Density greatly improved over SRAM
- ❑ Constant updates required to keep cell charges
- ❑ Complex addressing and refresh modes increase SEFIs

DRAM versus SRAM



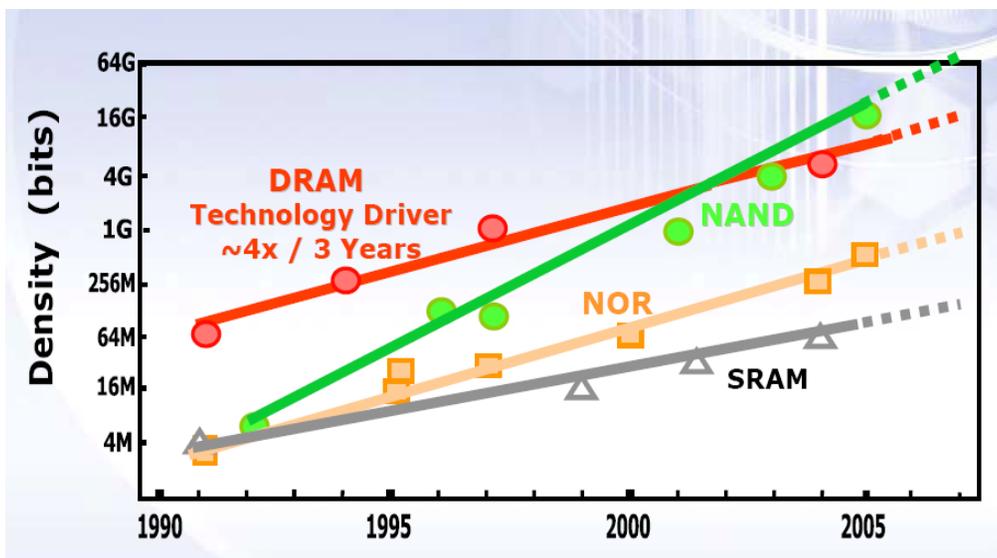
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- ❑ **Density, cost, latency, power are drivers**
 - **DRAM: greater density and reduced \$ per bit**
 - **SRAM: reduced latency, power and density**
- ❑ **Primary market driver is commodity PCs**
- ❑ **DRAM is typically preferred when density is the primary consideration (e.g. when size and weight are important) and SRAM is preferred when reducing data access latency is the primary concern**
- ❑ **The advent of synchronous and double data rate options have improved DRAM performance but has increased device access complexity and SEFI and SET effects**

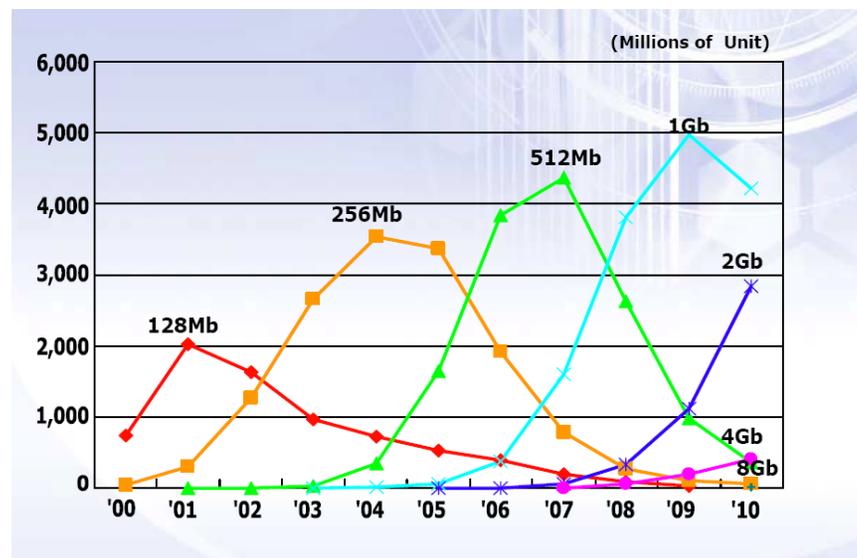


c/o ArchMemory.com

DRAM Trends



c/o Samsung



Source: Gartner Dataquest (May'05)

c/o Samsung

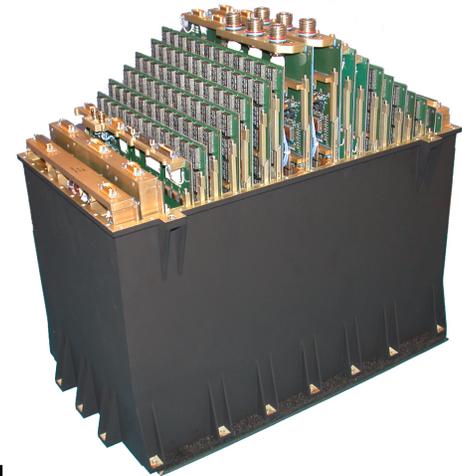
- ❑ Technology trends continue to provide higher density components
- ❑ Processing node scaling tends to improve TID but worsen SEE
- ❑ Obsolescence a key concern for aerospace systems
- ❑ Die revision changes within each family of devices another concern

Commercial DRAM in Space



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- ❑ **Commercial densities, performance and cost far outpacing rad-hard technologies and are being leveraged to produce increasingly capable systems**
- ❑ **For example, SDRAM used on SEAKR SSR with 2Tb capacity for a commercial imaging space system due to capture, compress and downlink data at rates up to 4Gbps**
 - **No other memory option could meet requirements**
- ❑ **Radiation effects such as TID and SEE need to be characterized and mitigated in systems using COTS memory for space**
 - **Telling that efforts to mitigate these effects more cost-effective than rad-hard for these missions**



DRAM On-orbit Analysis



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- ❑ While densities above 2Gb are now available in commercial components, lower density parts have been in orbit long enough to gather statistically significant data
- ❑ Recent study examined 64Mb, 128Mb, and 256Mb DRAM upset rates and compares them to predictions made using CREME96 (using actual space weather)
- ❑ On-orbit data collected from commercial DRAM parts used in SSRs designed and built by SEAKR Engineering was included in the study
- ❑ Study results presented at IEEE Aerospace Conference in March of 2009 [4]

DRAM Devices Studied



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❑ Samsung 64Mb KM44V16004 FPM CMOS DRAM

- Fast Page Mode (FPM) offers high speed random access of memory cells within the same row
- Revisions A, B, and D of this part used in the analyzed systems
- Rev. A has a 0.32 μm minimum feature size
- Rev. B & D have a 0.28 μm minimum feature size

❑ Samsung 128Mb KM44S32030 SDRAM

- Four die per package with a bit width of four
- Rev. B has a 0.20 μm minimum feature size

❑ Hitachi 256Mb HM5225405B SDRAM

- Four die per package with a bit width of four
- This component has a 0.16 μm minimum feature size

On-orbit DRAM Data



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| Density | Parts | Bits | Device hours |
|---------------|-------------|---------------|--------------|
| 64 Mb | 1155 | 74 Gb | 4.9E7 |
| 128 Mb | 1978 | 253 Gb | 1.1E8 |
| 256 Mb | 720 | 180 Gb | 2.0E7 |
| Totals | 3853 | 507 Gb | 1.8E8 |

□ Due to the large number of parts and distribution of varieties, this study believed to compile the most comprehensive on-orbit datasets of high density DRAM available

SSR Mission Descriptions

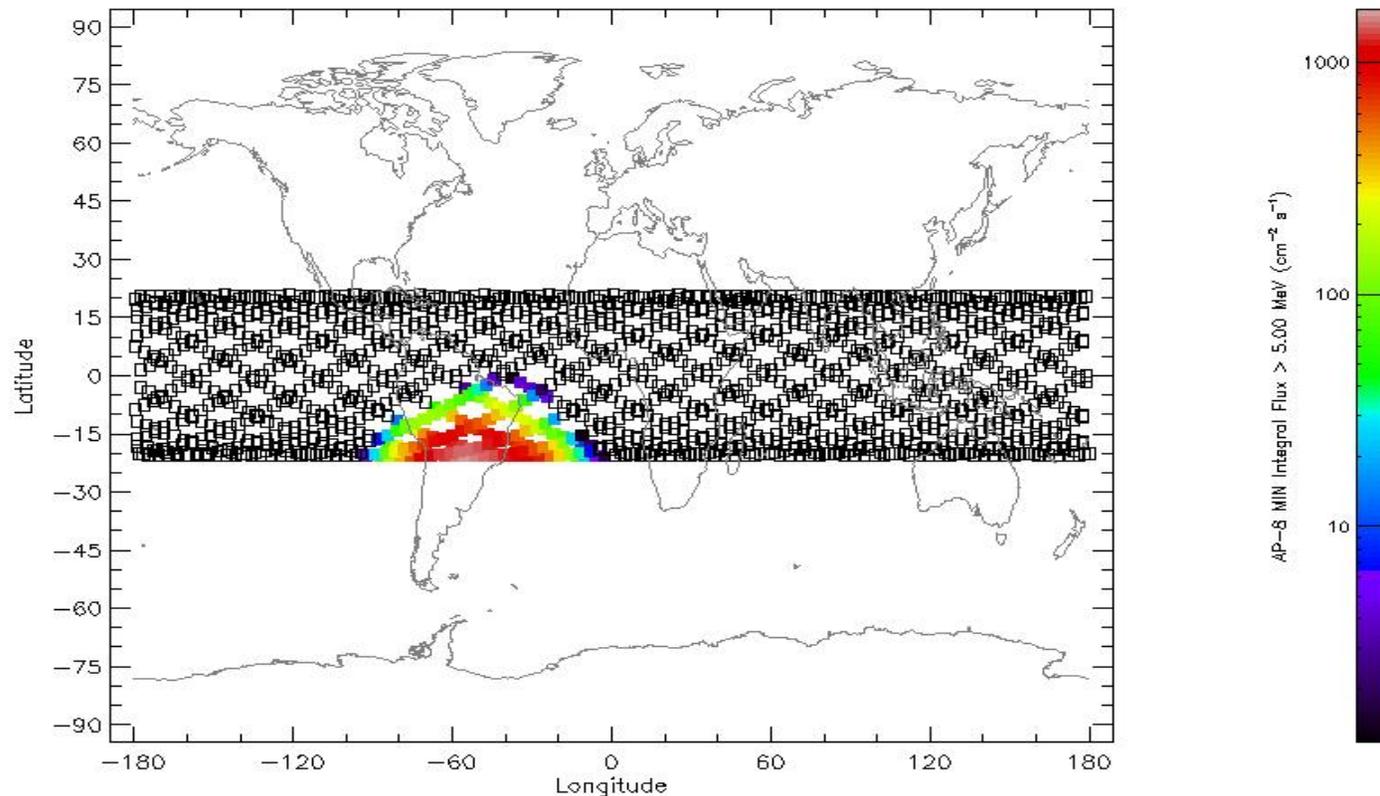


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| Memory Size | Manufacturer | Description | Program |
|-------------|--------------|---------------------------|-----------|
| 64Mb | Samsung | 16Mx4, 60nS, rev A | II,III,IV |
| 64Mb | Samsung | 16Mx4, 60nS, rev B | IV |
| 64Mb | Samsung | 16Mx4, 60nS, rev D | I |
| 128Mb | Samsung | 32Mx4, Synchronous, rev B | IV |
| 256 Mb | Hitachi | 64Mx4, Synchronous, 75ns | MRO |

- ❑ Missions I, II and III were various LEO orbits
- ❑ Mission IV was in the ISS orbit
- ❑ Data from the NASA's Mars Reconnaissance Orbiter (MRO) included as well

Mission I Orbit

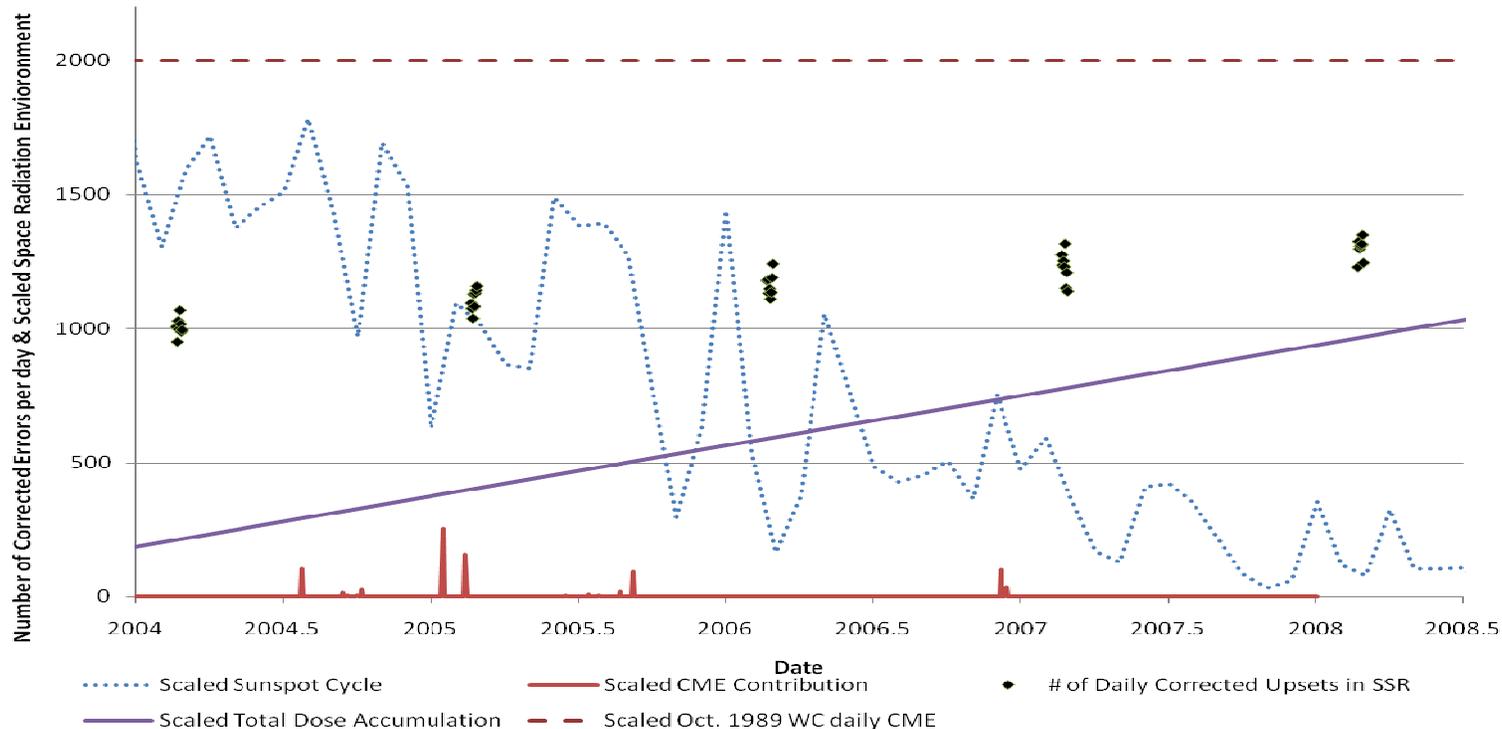


- ❑ Trapped protons along the orbit of Mission I showing multiple transits through the SAA
- ❑ Plot generated with SPENVIS

Mission II Results



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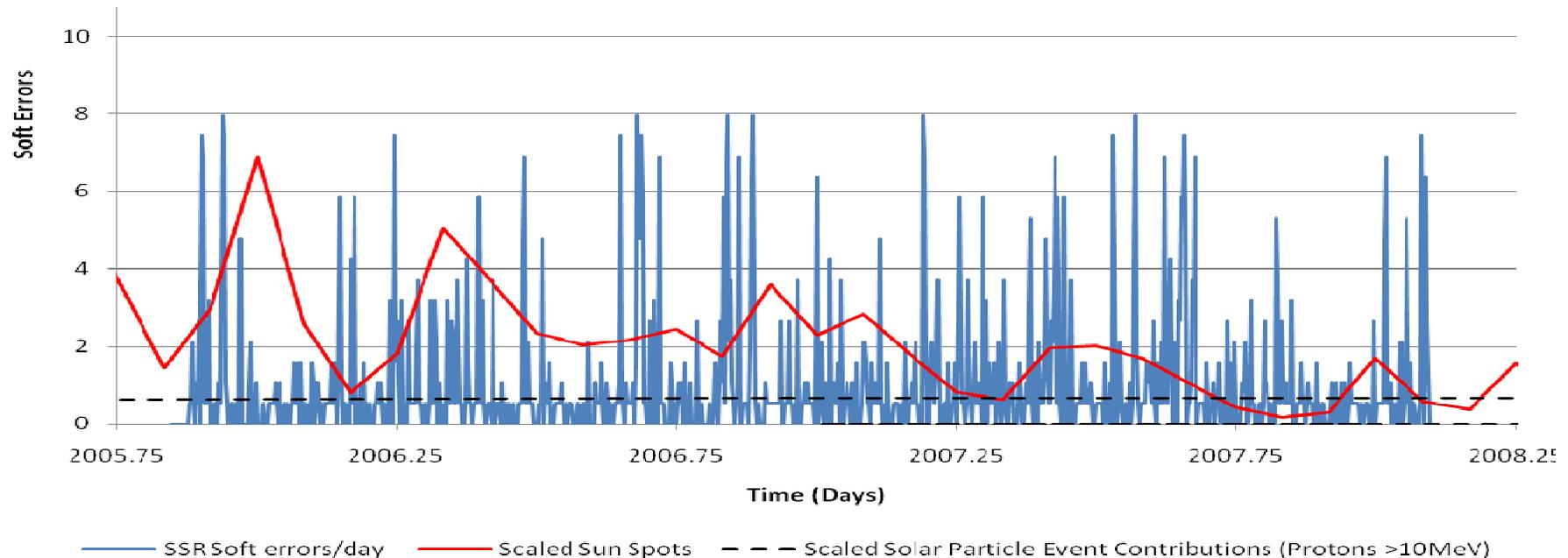


- ❑ Plot of several annual samples of daily upset data
- ❑ Scaled version of the sunspot cycle, relevant CMEs, and scaled accumulated TID overlaid
- ❑ Potential contributions to the linearly increasing number of upsets (all corrected by EDAC)

MRO Analysis Results



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- ❑ **SSR errors (all corrected by EDAC) have been plotted against the scaled sun spot data and CME data**
 - **No notable CME contributions and no substantial correlation between sunspots and errors**
 - **Errors occasionally correspond to periods of increased sunspot activity but no direct correlation**

Tool Assumptions



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| Program | Memory | GCR HI | GCR P | SEP HI | SEP P | AVG. TRP P |
|---------|------------------|--------|--------|--------|-------|------------|
| I | Samsung 64Mb | 99.98% | 99.98% | 0.02% | 0.02% | 100.00% |
| II | Samsung 64Mb | 99.91% | 99.91% | 0.09% | 0.09% | 100.00% |
| III | Samsung 64Mb | 99.99% | 99.99% | 0.01% | 0.01% | 100.00% |
| IV | Samsung 128Mb | 99.94% | 99.94% | 0.06% | 0.06% | 100.00% |
| V | Hitachi 256Mb | 99.99% | 99.99% | 0.01% | 0.01% | 0.00% |

Percentage of actual space environment contribution to upset rate in CREME96

Study Results



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| Program | Memory Type | On Orbit Upsets (Correctable Errors) per bit-day | Predicted CREME96 Upset Rate per bit-day | Ratio of Predicted CREME96 to On Orbit Upsets | Shielding Assumption (mils Al) | Orbit |
|---------|----------------|--|--|---|--------------------------------|-------|
| I | Samsung 64 Mb | 1.55E-9 | 1.58E-8 | 10.20 | 150 | LEO |
| II | Samsung 64 Mb | 4.44E-8 | 5.60E-8 | 1.26 | 150 | LEO |
| III | Samsung 64 Mb | 2.93E-8 | 3.15E-8 | 1.08 | 125 | LEO |
| IV | Samsung 128 Mb | 2.15E-10 | 1.23E-9 | 5.72 | 150 | ISS |
| MRO | Hitachi 256 Mb | 1.02E-11 | 6.29E-11 | 6.16 | 150 | Mars |

- ❑ Historical data for space weather which occurred during the missions used for the comparison
- ❑ Analysis assumed sensitive volume depth of 0.5µm

Tool Conservatism



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| Mils Aluminum | Program I | Program II | Program III |
|---------------|-----------|------------|-------------|
| 150 | 10.20 | 1.26 | 1.05 |
| 200 | 9.85 | 1.20 | 1.01 |
| 300 | 9.37 | 1.13 | 0.95 |

❑ Ratio of CREME96 predicted to actual upsets as a function of shielding thickness explored

- Other thicknesses explored ranging from 150 mils to 300 mils
- Less than a 17% change was observed in the predicted upset rate
- Increased shielding leads to rates slightly closer to on-orbit rates
- Sensitive volume (SV) assumptions also explored
 - SV effects for Earth-orbiting spacecraft at LEO negligible compared to the trapped proton contributions
- Minor difference between observed and predicted upsets indicates models relatively accurate in predictions
- Ongoing research to explore additional possibilities
- No one size fits all for guessing tool conservatism

Summary of Characteristics



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| Type | Volatile | Read Latency | Write Latency | Lifespan | Power Usage | Capacity | TID |
|--------|----------|--------------|---------------|----------|-------------|----------|--------|
| EEPROM | NV | ~200ns | ~3ms | ~1e6 | Low | Low | High |
| Flash | NV | ~70ns | ~500ms | ~1e5 | Low | High | Low |
| SRAM | V | ~10ns | ~10ns | Infinite | Medium | Low | High |
| DRAM | V | ~50ns | ~50ns | Infinite | High | Medium | Medium |

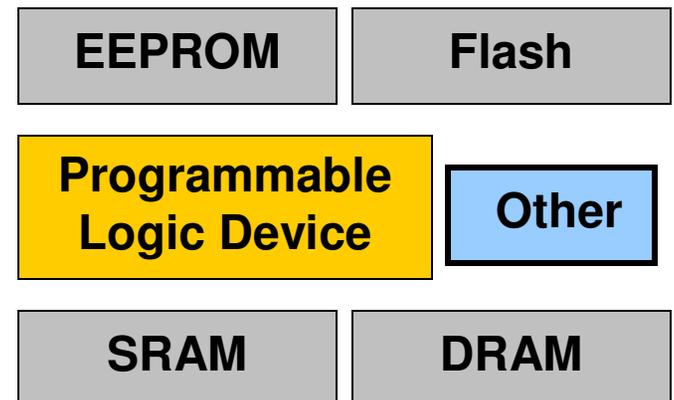
- ❑ **Volatile memory provides advantages of reduced access latency and extended device life spans but at the cost of power and data volatility**
- ❑ **Non-volatile memory has opposite characteristics**
- ❑ **Flash and DRAM principally from commercial market**

Other Memory Types



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- ❑ Efforts underway to develop an alternative means of storage than the tried and true electrons and charge
- ❑ Developers seeking to create a “universal memory”
 - Non volatile with good density and fast access
- ❑ Promising options include
 - Ferroelectric RAM
 - Magnetoresistive RAM
 - Chalcogenide RAM



- ❑ Memory with other types of applications are also being explored for space systems

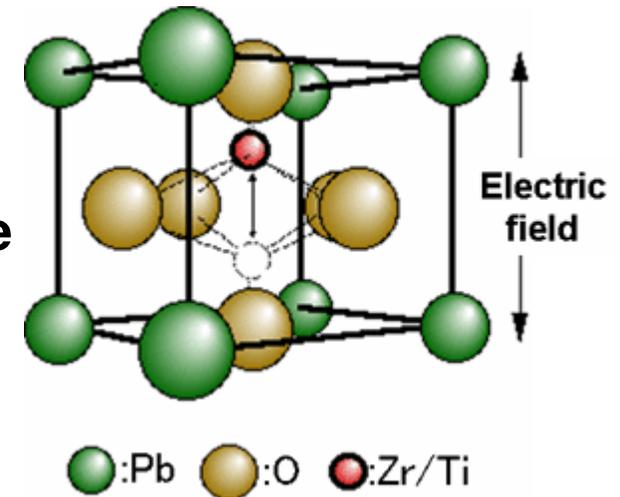
FRAM Structure



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❑ Ferroelectric Random Access Memory

- Ferroelectric film (capacitor) for storing data
- Electric polarization of atom determines value
- Low power required to change value
 - No charge pump required
- No charge required to maintain persistence



c/o Fujitsu

❑ Characteristics

- Non-volatile memory with high endurance
- High-speed access, low power consumption
- Characteristics split the difference between RAM and ROM
- Marketed for smart cards and hand-held devices

❑ Options: Fujitsu, Ramtron (under development)

MRAM Structure

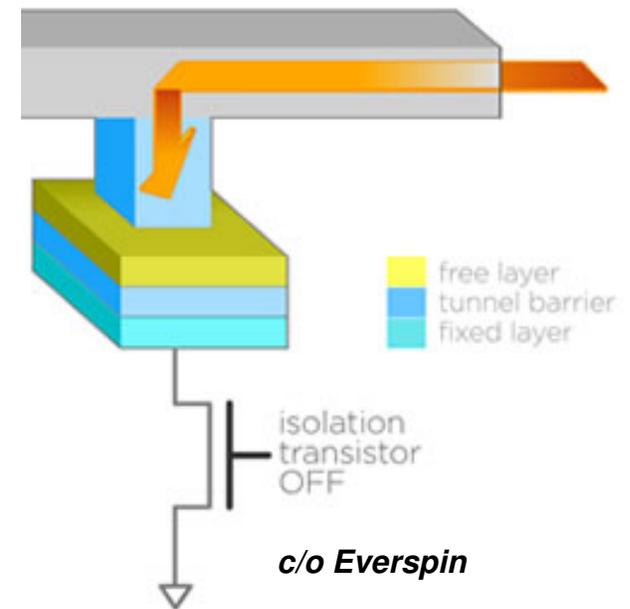
❑ Magnetoresistive Random Access Memory

- Magnetic Tunnel Junction (MTJ) stores data
- Magnetic polarization determines electric resistance which is sensed to read value
- Low power required to change value
 - No charge pump required
- No charge required to maintain persistence

❑ Characteristics

- Non-volatile memory with high endurance
- High-speed access, low power consumption
- Characteristics split the difference between RAM and ROM
- Strong commercial support to create the “universal memory”

❑ Options: Everspin, Hitachi, IBM, NEC, Toshiba



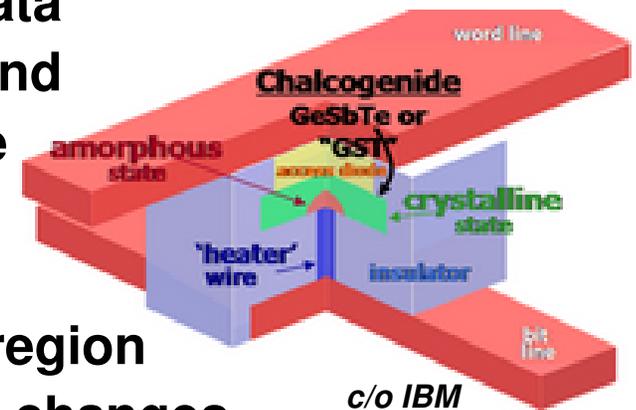
C-RAM Structure



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❑ Chalcogenide Random Access Memory

- Heat applied to Chalcogenide glass stores data
- Crystalline and amorphous states possible and resistance change is sensed to read value
 - Optical properties of material used in CD/DVDs
- No charge required to maintain persistence
- High programming current density in active region
- Temperature sensitivity may require process changes
- Also known as Phase-change memory



❑ Characteristics

- Non-volatile memory with high endurance
- High-speed access, low power consumption
- Characteristics split the difference between RAM and ROM

❑ Options: BAE, Hitachi, IBM, Intel, Samsung, etc.

- BAE: 4Mb, TID to 1 Mrad with 70ns read and 500ns write latency

Conclusions



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- ❑ **Four standard memory types examined and contrasted**
- ❑ **DRAM study of on-orbit SEAKR SSRs outlined**
 - **On-orbit upset data from 3853 COTS DRAM components**
 - **Included 64, 128, and 256Mb densities**
 - **Representing ~180 million on-orbit device hours**
 - **CREME96 predictions, using observed space weather, shown to be up to 10 times more conservative than observed upsets**
 - **No single attribution is clear in the analysis at present**
- ❑ **Alternative memory options examined**
- ❑ **Commercial devices currently provide improved performance per cost for space systems, even including radiation testing and mitigation**
- ❑ **Hopefully commercial trends continue in our favor**

References



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- ❑ [1] Gregory R. Allen, “Compendium of Test Results of Single Event Effects Conducted by the Jet Propulsion Laboratory,” *Proc. IEEE Nuclear and Space Radiation Effects Conference (NSREC) Radiation Effects Data Workshop*, Tucson, AZ, July 14-18, 2008.
- ❑ [2] Toshiba, Inc., “NAND vs. NOR Flash Memory Technology Overview,” Company Whitepaper, Irvine, CA, April 5, 2006.
- ❑ [3] T. Oldham, M. Suhail, M. Friendlich, et al., “TID and SEE Response of Advanced 4G NAND Flash Memories,” *Proc. IEEE Nuclear and Space Radiation Effects Conference (NSREC) Radiation Effects Data Workshop*, Honolulu, HI, July 23-27, 2007.
- ❑ [4] Ian Troxel, Chris Miller, Russ Owen, et al., “Trends in Radiation Susceptibility of Commercial DRAMs for Space Systems,” *Proc. IEEE Aerospace Conference*, Big Sky, MT, March 9-13, 2009.
- ❑ ... and numerous vendor web sites

Contact Information



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Questions?